

# Design of GNSS Aided Inertial Navigation System

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# Introduction

- GNSS-aided INS provides accurate and timely location information.
- The project allows sea, land, and air vehicles to find their positions with low error.
- The unit consist of 3 main parts; a processor unit, an IMU sensor, and GPS receiver sensor.
- Processor unit finds the distance and bearing angle from IMU sensor and coordinates from GPS, use this data to find location by reducing error.

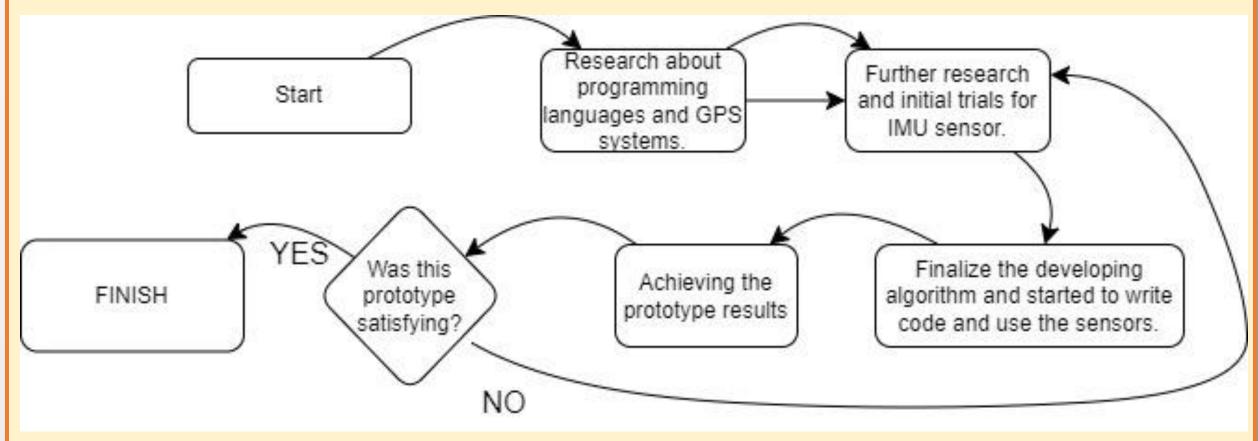
#### **Design Process**

# **Application Areas**

GNSS systems are used in ground, air and land vehicles.
The systems are used in military applications, cargo ships, submarines, automobiles, transport fleets, airplanes, helicopters.



- Location finder designs are all about providing solutions to realtime problems.
- Since the selected products must be cost-efficient, the algorithm should work to maximize efficiency.



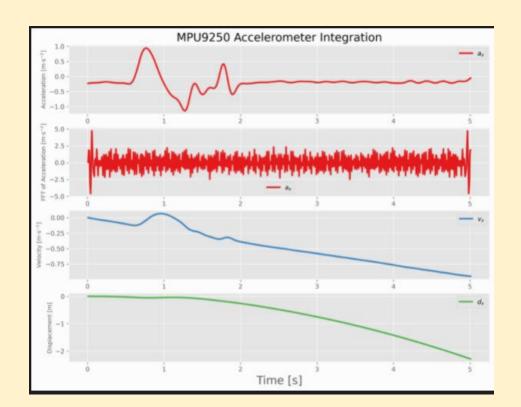
Focused on optimizing algorithms for efficiency to maximize the performance of the location finder while minimizing computational resources.

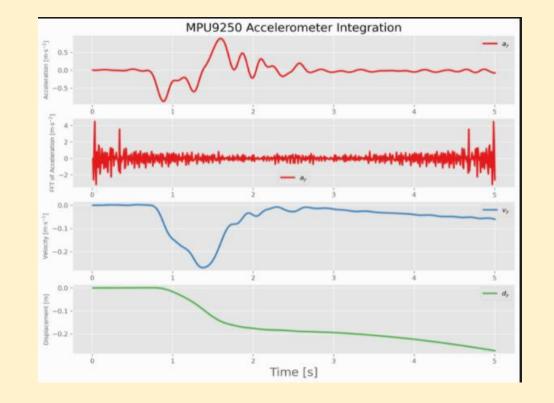
### **Solution Methodology**

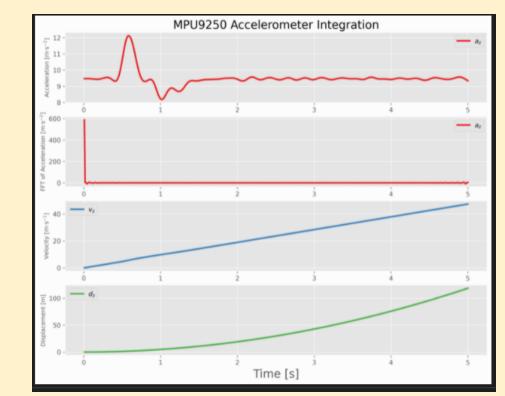
In order to convert device-referenced acceleration data to

#### **Results and Discussion**

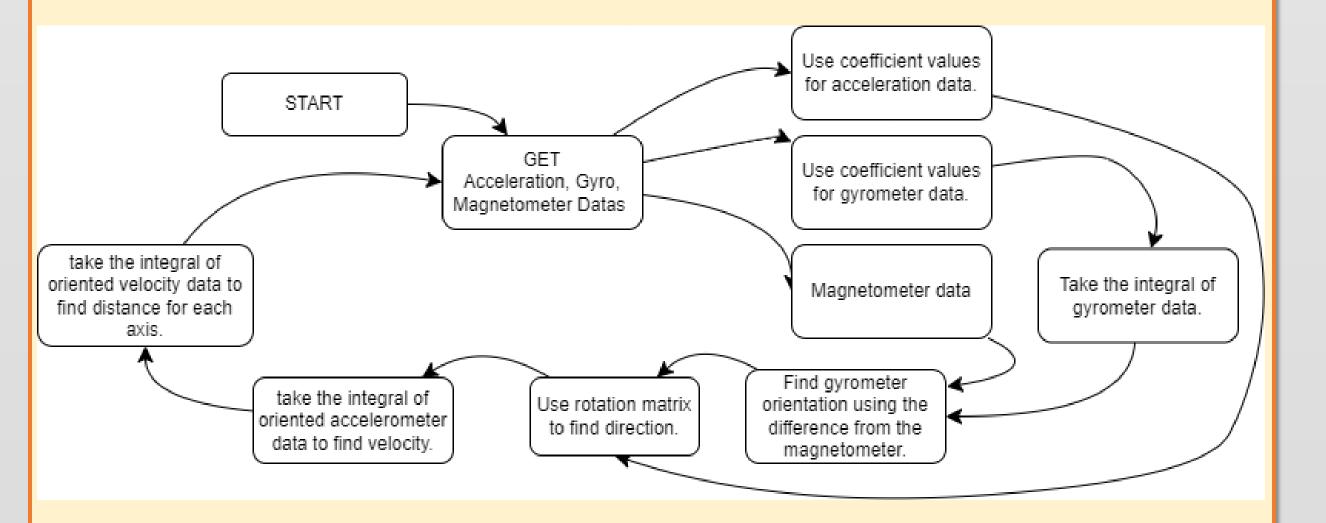
The images below show graphs related to acquiring accelaration data, applying rotation matrix and filtering







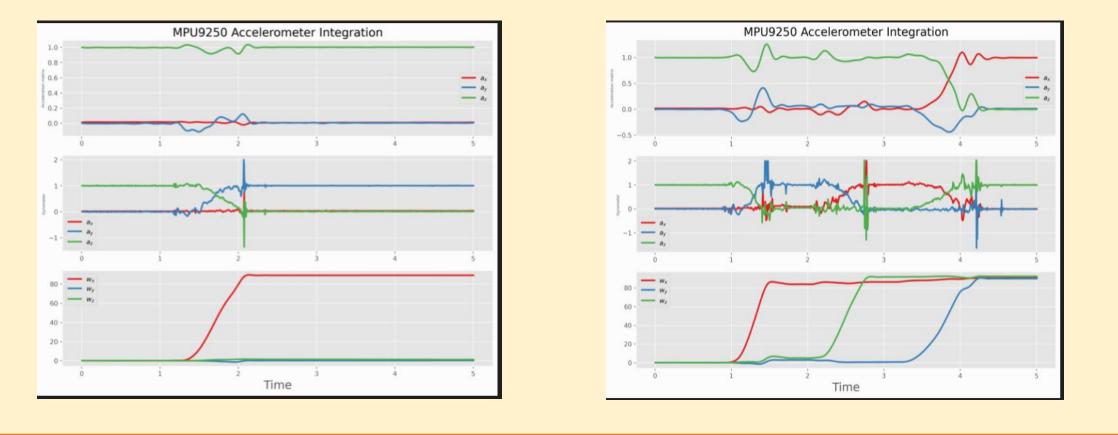
distance, the integral of the received data should be taken 2 times in the interval t. The shorter the duration t, greater the accuracy of the distance from the inertial sensor.



The rotation matrix is used to convert Eart-referenced acceleration data into device-referenced acceleration data.

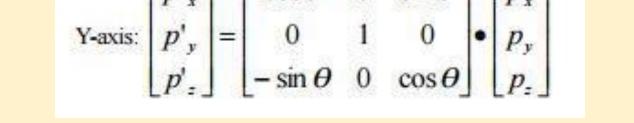
Z-axis:	$p'_{v}$	=	$\sin \theta$	$\cos\theta$	0 •	$p_{v}$
	p'.		0	$-\sin\theta$ $\cos\theta$ 0	1	$p_{z}$
	$p'_x$	] [	1 (	) 0	ſ	$\int p_x$
X-axis:	$p'_y$	=	0 cos	$s\theta - si$	n <i>θ</i> •	$p_y$
	p':.		0 sin	$\theta$ cos	θ	<i>p</i> :

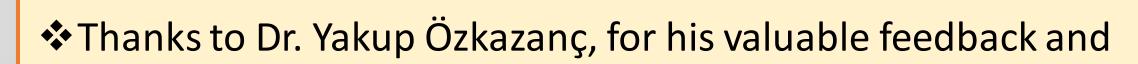
#### The images below show 90° rotation in the roll, pitch, yaw axes.



#### Acknowledgements

This project was completed within the context of ELE401-401 Graduation Project courses in Hacettepe University, Faculty of Engineering, Department of Electrical and Electronics Engineering.





#### guidence during the development of this project.